#### Recap

We had introduced Euler's axioms where we found that we needed to find the net external forces and net ext. moments acting on a system. Towards this goal, we had introduced the idea of equivalent force systems, using which we reduced the effect of several forces and couples acting on a system to an equivalent single resultant force and a single resultant couple, or an equivalent wrench.

Before we find the equivalent force system, it is needed that we identify all the forces and couples that are exerted on the system — the external forces and couples. In the last lecture, we started discussing how to draw FBDs where we identify all contact and body forces and contact moments acting externally to the system under consideration.

In this lecture, we will look at a class of ext. contact forces & moments that are generated (than directly applied) when a system's movement is constrained by some supports. Called reactions

#### Reactions at Supports

In many situations, the motion of an RB or a system is constrained by some supports



When a force is applied on the system, the support generates a force system (force and moment/couple)

contact force & couple The surface force system that develop at the supports or points of contact on the body of interest is called the reaction force system (in short REACTIONS)

While drawing an FBD of a system, we should remove the physical details of the support and replace those details by a reaction force system!



General Rule: If the support prevents translation in a given direction, then a force must be developed on the system in that direction. Likewise, if rotation is prevented, a couple moment must be exerted on the system.



Let us consider few cases of different support systems in AD first and examine the reactions generated.

# Support reactions for aD analysis





## Reaction force systems in 3D analysis



b> how to replace a force system by an equivalent resultant force system at a given point

These will be useful when we deal with Euler's axioms Recall the two axioms:

$$\frac{|\text{st axiom}:}{P|_{I}} = F_{R} \qquad (\text{Rate of change of linear} \\ \text{momentum} = \text{net external} \\ \Rightarrow m \underline{a}_{C|I} = F_{R} \\ \text{force})$$

$$\frac{\text{renter of mass (COM)}}{\text{center of mass (COM)}}$$

$$\frac{\text{Rate of change of angular}}{\text{momentum} = \text{net external}} \\ \text{point O} \\ \text{must be a point} \\ \text{fixed in the inertial}} \qquad [Note: Forces & moments]$$

These two sets of relations leads to the equations of motion

which are effectively six scalar equations (equivalent to two vector equations of motion). Together with appropriate <u>initial</u> <u>conditions</u>, these are used to determine the general motion of an RB.

Determination of this motion, however, does not always split neatly into a translational part and a rotational part. Infact in many cases, these two parts are coupled together that give rise to strongly coupled system of nonlinear ODEs

Nonlinear  
ODE: 
$$\frac{d \times (t)}{dt} = \frac{g}{2} \left( \times (t), F_R(t), M_0(t) \right)$$
  
Initial condition:  $\times (t=0) = \times_0$   
 $x = \begin{bmatrix} \times tran \\ \times rot \end{bmatrix}$ 

Euler's 2nd axiom for a Moving Reference point

We turn to Euler's and axiom:  $\underbrace{Ho}_{II} = \underline{M}_{O}$ and recall that this relation holds only for a point O fixed in the inertial frame I.



Given any general point 'A' moving w.r.t. I (see above figure) Is  $\dot{H}_{A|I} = M_A$  VALID? If so, under what circumstances?

To derive an alternate form of Euler's and axiom valid for a moving point and to determine any restrictions on its use, lets first derive a relationship for transfer of angular mamentum from one pt to another and then we will derive a relationship between the rate of change of angular momentum about a point A and the net external moment abt A.

### Transformation rule for Angular Momentum

Angular momentum of an RB abt a moving pt A was defined earlier:

$$H_{AIF} = \int Y_{PA} \times Y_{PA|F} dm$$

$$C: Center of mass of RB$$

$$Et can be shown that$$

$$H_{AIF} = H_{CIF} + m(Y_{CA} \times Y_{CAIF}) \qquad (Proved next)$$
angular mom. total mass of RB

Proof of 
$$\underline{H}_{A|F} = \underline{H}_{C|F} + m(\underline{\Upsilon}_{CA} \times \underline{\Upsilon}_{CA|F})$$

Start with the defn of angular  
momentum of RB about moving pt A  

$$H_{A|F} = \int (\Upsilon_{PA} \times \Upsilon_{PA|F}) dm$$

$$= \int (\Upsilon_{Pc} + \Upsilon_{CA}) \times \Upsilon_{PA|F} dm$$

$$= \int (\Upsilon_{Pc} + \Upsilon_{CA}) \times (\Upsilon_{Pc]F} + \Upsilon_{CA|F}) dm$$

$$V_{P}|F = \Upsilon_{P|F} - \Upsilon_{A|F}$$

$$= \int (\Upsilon_{Pc} \times \Upsilon_{Pc|F}) dm + \int (\Upsilon_{CA} \times \Upsilon_{Pc|F}) dm$$

$$= \Upsilon_{P|F} - \Upsilon_{c|F}$$

+ 
$$\int (\underline{\Upsilon}_{PC} \times \underline{\vee}_{CA|F}) dm + \int (\underline{\Upsilon}_{CA} \times \underline{\vee}_{CA|F}) dm$$

# Simplify

Ξ

Term 
$$\int (\underline{x}_{PC} \times \underline{x}_{CA|F}) dm = (\int \underline{x}_{PC} dm) \times \underline{x}_{CA|F}$$
 the integ.  
 $\underline{x}_{PC} = \underline{x}_{PO} - \underline{x}_{CO}$ 

$$= \int (\underline{x}_{PO} - \underline{x}_{CO}) dm \times \underline{x}_{CA|F}$$

$$= (\int \underline{x}_{P} dm - \underline{x}_{C}) dm \times \underline{x}_{CA|F}$$

$$= (\int \underline{x}_{P} dm - \underline{x}_{C}) dm \times \underline{x}_{CA|F}$$

$$= (\underline{m}\underline{x}_{C} - \underline{m}\underline{x}_{C}) \times \underline{x}_{CA|F}$$

$$= 0$$
Term  $\underline{x}$ 

$$\int (\underline{x}_{CA} \times \underline{x}_{CA|F}) dm = (\underline{x}_{CA} \times \underline{x}_{CA|F}) \int dm$$

$$= 0$$

$$\frac{de_{T}}{de_{T}} dm = (\underline{x}_{CA} \times \underline{x}_{CA|F}) \int dm$$

$$\frac{de_{T}}{de_{T}} dm = (\underline{x}_{CA} \times \underline{x}_{CA|F}) \int dm$$

 $= m \underline{\Upsilon}_{CA} \times \underline{\nabla}_{CA|F}$ 

t

Therefore,

 $\underline{H}_{A|F} = \underline{H}_{C|F} + m \underline{\Upsilon}_{CA} \times \underline{\nabla}_{CA|F}$ 

The above angular momentum transfer rule is valid for any reference frame F, and therefore is also valid for inertial frames of reference. Next, we intend to find a relationship between  $\frac{H}{AII} = \frac{d}{dt} \left\{ H_{AII} \right\}_{I} \xrightarrow{??} M_{A}$ 

# HAII - MA relationship



Using angular momentum transfer sule, we can write:

$$\square \quad \exists_{AII} = \exists_{CII} + m(\Upsilon_{CA} \times \Upsilon_{CAII})$$

2 
$$\underline{H}_{O|I} = \underline{H}_{C|I} + m(\underline{Y}_{CO} \times \underline{Y}_{CO|I})$$

Substitute Eq. (2) in Eq. (1):

$$\underline{H}_{A1E} = \left[\underline{H}_{01E} - m\left(\underline{r}_{CO} \times \underline{V}_{CO|E}\right)\right] + m\left(\underline{r}_{CA} \times \underline{V}_{CA|E}\right)$$

Differentiate both sides wit time in the 'I' frame:

$$\dot{H}_{A1I} = \frac{d}{dt} \left\{ \left[ \frac{H_{01I} - m \left( \underline{r}_{CO} \times \underline{v}_{CO|I} \right) \right] + m \left( \underline{r}_{CA} \times \underline{v}_{CA|I} \right) \right\} \right|_{I}$$

$$\frac{d}{dt} \left\{ \underbrace{H}_{0|L} \right\}_{L} = \underbrace{H}_{0|L} = \underbrace{M}_{0} \leftarrow \text{Net moment due to all ext forces} \\ \text{and couples about 0, 0 \in I} \\ \frac{d}{dt} \left( \underline{Y}_{C0} \times \underline{Y}_{C0|L} \right) \Big|_{L} = \frac{d}{dt} \left\{ \underline{Y}_{C0} \right\}_{L} \times \underline{Y}_{C0|L} + \underline{Y}_{C0} \times \underline{d}_{dt} \left\{ \underline{Y}_{C0|L} \right\}_{L} \\ = \underbrace{Y}_{C0|L} \times \underline{Y}_{C0|L} + \underbrace{Y}_{C0} \times \underline{Q}_{C0|L} \\ \frac{d}{dt} \left( \underline{Y}_{CA} \times \underline{Y}_{CA|L} \right) \Big|_{L} = \frac{d}{dt} \left\{ \underline{Y}_{CA} \right\}_{L} \times \underline{Y}_{CA|L} + \underbrace{Y}_{CA} \times \underline{d}_{dt} \left\{ \underline{Y}_{CA|L} \right\}_{L} \\ = \underbrace{Y}_{CA|L} \times \underline{Y}_{CA|L} + \underbrace{Y}_{CA} \times \underline{Q}_{CA|L} \\ \frac{d}{dt} \left( \underline{Y}_{CA} \times \underline{Y}_{CA|L} \right) \Big|_{L} = \frac{d}{dt} \left\{ \underline{Y}_{CA} \right\}_{L} + \underbrace{Y}_{CA} \times \underline{Q}_{CA|L} \\ = \underbrace{Y}_{CA|L} \times \underbrace{Y}_{CA|L} + \underbrace{Y}_{CA} \times \underline{Q}_{CA|L} \\ \frac{d}{Q} + \underbrace{Y}_{CA} + \underbrace{Q}_{CA|L} + \underbrace{Y}_{CA} \times \underline{Q}_{CA|L} \\ \frac{d}{Q} + \underbrace{Y}_{CA} + \underbrace{Q}_{CA|L} + \underbrace{Y}_{CA} \times \underline{Q}_{CA|L} \\ \frac{d}{Q} + \underbrace{Y}_{CA} + \underbrace{Q}_{CA|L} + \underbrace{Y}_{CA} \times \underline{Q}_{CA|L} \\ \frac{d}{Q} + \underbrace{Y}_{CA} + \underbrace{Q}_{CA|L} + \underbrace{Y}_{CA} \times \underline{Q}_{CA|L} \\ \frac{d}{Q} + \underbrace{Y}_{CA} + \underbrace{Q}_{CA|L} + \underbrace{Y}_{CA} \times \underline{Q}_{CA|L} \\ \frac{d}{Q} + \underbrace{Y}_{CA} + \underbrace{Q}_{CA|L} + \underbrace{Y}_{CA} \times \underline{Q}_{CA|L} \\ \frac{d}{Q} + \underbrace{Y}_{CA} + \underbrace{Q}_{CA} + \underbrace$$

$$\Rightarrow \overset{\cdot}{H}_{A \mid I} = \overset{M}{H}_{0} - (\overset{\Gamma}{I}_{A} \times \overset{K}{F}_{R}) - (\overset{\Gamma}{Y}_{CA} \times \overset{m}{M} \overset{Q}{A}_{A \mid I})$$
Recall
from Lec 8:
$$\overset{M}{H}_{0} = \overset{M}{M}_{A} + \overset{K}{Y}_{Ao} \times \overset{F}{F}_{R} = \overset{M}{M}_{A} + \overset{K}{Y}_{A} \times \overset{F}{F}_{R}$$

$$\overset{M}{H}_{R} = \overset{N}{\underset{i=1}{\sum}} \overset{T}{I}_{iB} \times \overset{F}{F}_{i} + \overset{T}{\underset{j=1}{\sum}} \overset{G}{G}_{j}$$

$$= \overset{N}{\underset{i=1}{\sum}} (\overset{K}{Y}_{AB} + \overset{K}{I}_{A}) \times \overset{F}{F}_{i} + \overset{T}{\underset{j=1}{\sum}} \overset{G}{G}_{j}$$

$$= \overset{N}{\underset{i=1}{\sum}} \overset{N}{Y}_{iA} \times \overset{F}{F}_{i} + \overset{T}{\underset{j=1}{\sum}} \overset{G}{G}_{j} + \overset{N}{\underset{i=1}{\sum}} \overset{K}{F}_{R} \times \overset{K}{F}_{R}$$

$$= \overset{N}{\underset{i=1}{\sum}} \overset{N}{Y}_{iA} \times \overset{F}{F}_{i} + \overset{T}{\underset{j=1}{\sum}} \overset{G}{G}_{j} + \overset{N}{\underset{i=1}{\sum}} \overset{K}{F}_{AB} \times \overset{K}{F}_{R}$$

$$\Rightarrow \overset{H}{H}_{A \mid I} = \overset{M}{M}_{A} - \overset{K}{Y}_{CA} \times \overset{m}{M}_{A \mid I} = \overset{Alternative form of Euler's and axiom for moving$$

Revisiting the question:  
Given any general point 'A' moving w.r.t. I  
Is 
$$\dot{H}_{A|I} = M_A$$
 VALID? If so, under what  
circumstances?

ref. point A



In this example, lets list out the points (P) for which  $\frac{H}{P}II = MP$  is valid!

Case I: 
$$V_0 = 0$$
 and  $a_0 = 0$  A, B, C, D, E

#### Reason

$A \rightarrow$	$\Delta_{A I} = Q$	Bω, ώ
$B \rightarrow$	Ύсв // ДвII	<u>A</u> BIT
$C \rightarrow$	$\underline{\Upsilon}_{cc} = \underline{O}$ (COM)	
$D \rightarrow$	$\Upsilon_{CD} // \ \Delta_{PII} = \omega^2 r \hat{e}_2$	$(\underline{\alpha}_{A I} = \underline{0})$ $E$ $\nabla_{o_1} \alpha_{o_2}$
$\vdash \rightarrow$	$\underline{\alpha}_{E I} = \underline{0}$	
		Points where valid
Case I	$v_0 \neq 0$ and $a_0 = 0$	A, B, C, D, E

Same reasons as case I

Points where valid



If  $\dot{w} = 0$ , then  $\underline{a}_{FII}$  becomes parallel to  $\underline{r}_{CF}$  and pt. F becomes a valid point! accelerating towards or away from COM

... pt F is valid when w = 0